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# Trajectory Tracking Control Design for Large-Scale Linear Dynamical Systems With Applications to Soft Robotics

Maxime Thieffry<sup>ID</sup>, Alexandre Kruszewski, Thierry-Marie Guerra, and Christian Duriez, *Senior Member, IEEE*

**Abstract**—This article presents new results to control process modeled through linear large-scale systems. Numerical methods are widely used to model physical systems, and the finite-element method is one of the most common methods. However, for this method to be precise, it requires a precise spatial mesh of the process. Large-scale dynamical systems arise from this spatial discretization. We propose a methodology to design an observer-based output feedback controller. First, a model reduction step is used to get a system of acceptable dimension. Based on this low-order system, two linear matrix inequality problems provide us, respectively, with the observer and controller gains. In both the cases, model and reduction errors are taken into account in the computations. This provides robustness with respect to the reduction step and guarantees the stability of the original large-scale system. Finally, the proposed method is applied to a physical setup—a soft robotics platform—to show its feasibility.

**Index Terms**—Large-scale systems, model-order reduction, robust control, soft robotics.

## I. INTRODUCTION

THE use of simulation shortens the development time of engineering systems. Different designs, actuation or sensing methods, and control approaches can be tested out quickly. For computer-based simulations, the demand for a highly accurate description of realistic phenomena leads to work with systems of very large dimensions. There are many

examples of large-scale systems that present challenges, such as power networks, urban traffic networks, digital communication networks, or robotics. When these systems are part of a control loop, one is interested in their reactions with respect to these inputs. This article is about designing a controller for such large systems.

Solutions have been proposed to control large-scale systems: for nonlinear systems, Tong *et al.* [1] and Hsiao *et al.* [2] present different methodologies to design controllers for nonlinear large-scale systems based on decentralized fuzzy control. Simulation results show the interest and the effectiveness of the approach, but examples are restricted to interconnected systems of small dimensions compared with examples available in the numerical simulations' literature.

A second track investigated to study the systems of large dimensions is model-order reduction. It provides a low-order model that represents the behavior of the large-scale system and it is more suitable to design a controller for this reduced-order model. However, due to the approximation errors, controllers designed for this low-order model can lead to unstable closed-loop systems when applied to the large-scale plants. To handle this issue, a method to build a low-dimension controller by optimizing the  $H_\infty$  norm of the low-order model is presented in [3]. This method ensures the stability of the large-scale closed-loop model. A similar idea is presented in [4], where a low-order controller is computed while guaranteeing the large-scale closed-loop stability through linear matrix inequality (LMI) constraints.

The theoretical results of this article aim at being generic regarding the range of applications. In this article, soft robotics examples illustrate the proposed method. Compliant bodies offer high adaptability to environment and inherent safe interactions with humans. In addition, soft robotics is an active field of research, especially for closed-loop dynamic control [5], [6] that makes this new type of robots a good experimental platform. When dealing with soft robots control, significant results exist for kinematics control, but a few methods exist for dynamic control [6]. Many works are focused on open-loop control of soft robots, such as [7] or [8]. The computational cost of accurate models is a major drawback of model-based controllers, both because of the complexity of the design and for the practical implementation on the hardware. To avoid this constraint, Thuruthel *et al.* [9] proposed a learning-based open-loop dynamic controller to perform dynamic motions. However, open-loop strategies suffer from well-known bottlenecks, such as the robustness of the controller with respect

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to modeling uncertainties and the response of the system with respect to external perturbations. This is why closed-loop algorithms are presented hereafter. A dynamic controller based on constant curvature assumptions is presented in [10], where the model is used to generate a feed-forward action coupled with a PID controller. Also based on the piecewise constant curvature model, Della Santina *et al.* [11] presented a dynamic controller that enables dynamic trajectory tracking for a continuous soft robot while handling interactions with the environment. Our approach aims at being generic regarding the geometry of the robot, and thus, we propose to design the control law based on a mechanical model of the structure.

A simulation framework to design, model, and control soft robots is presented in [12]. It uses the finite-element method (FEM) to simulate soft bodies in real time and it has been used to propose different methods to control soft robots. Coevoet *et al.* [13] presented an open-loop optimization-based method to control soft robots interacting with its environment. Then, [14] proposes a closed-loop inverse kinematics controller based on a quasi-static FEM model. To overcome this quasi-static assumption, [15] presents a model-based dynamic controller for soft robots. This article proposes to use FEM to build a large-scale dynamic model of the system and proposes a methodology to perform dynamic trajectory tracking.

#### A. Notations and Definitions

The notations used in this article are standard. We denote that  $\mathbb{R}$  is the set of real numbers,  $\mathbb{R}^n$  is the set of real vectors of dimension  $n$ , and  $\mathbb{R}^{n \times m}$  is the set of matrices with  $n$  rows and  $m$  columns. For any vector or matrices  $x$  or  $M$  in  $\mathbb{R}^n$ ,  $\mathbb{R}^{n \times m}$ ,  $x^T$ , and  $M^T$  are their transpose. For any matrix  $M \in \mathbb{R}^{n \times n}$ ,  $M = M^T > 0$  defines a symmetric positive definite matrix. For a vector  $x \in \mathbb{R}^n$ , the notation **diag**( $x$ ) refers to a square diagonal matrix, where the elements of the diagonal are the elements of  $x$  and the entries outside the diagonal are all zeros.

The time dependence of velocity, displacement, state, and input vectors  $v(t)$ ,  $q(t)$ ,  $x(t)$ , and  $u(t)$  is omitted. This article is concerned with discrete-time systems and the value of any variable  $x$  at time  $t_k$  is simply written as  $x = x(t_k)$  and the value of this variable at time  $t_{k+1}$ , i.e., at the next time step, is written as  $x_+ = x(t_{k+1})$ .

Given a vector  $x \in \mathbb{R}^n$  and function  $V(x) : \mathbb{R}^n \mapsto \mathbb{R}$ , we denote  $\Delta V(x) = V(x_+) - V(x)$ . A star  $*$  indicates a transposed quantity in an expression or in a symmetric matrix. For example,  $*PA - P < 0$  stands for  $A^T P A - P < 0$  and  $(\begin{smallmatrix} P & * \\ A & P \end{smallmatrix}) < 0$  stands for  $(\begin{smallmatrix} P & A^T \\ A & P \end{smallmatrix}) < 0$ .

## II. PROBLEM STATEMENT

### A. Introduction

Let us start from a linear large-scale system given in the following equation:

$$M : \begin{cases} x_+ = Ax + Bu \\ y = Cx \end{cases} \quad (1)$$

where  $A \in \mathbb{R}^{n \times n}$ ,  $B \in \mathbb{R}^{n \times m}$ ,  $m$  being the number of actuators,  $C \in \mathbb{R}^{p \times n}$  is a matrix defining the system's outputs, and  $p$  is the number of outputs.

System matrices  $A$ ,  $B$ , and  $C$  come from numerical methods, such as the FEM. For this model to be precise, the number of variables in the state vector  $x$  has to be significantly high and system (1) is thus a large-scale system with  $n \gg 1$ .

The large dimension of system (1) makes difficult the use of standard tools of automatic control (such as pole placement or LMI constraints problem) to design a controller for this system. To tackle this problem, we propose to use model-order reduction methods to obtain a low-order system that represents faithfully the full-order system. Based on this low-order state, the designs of both the controller and the observer are tractable.

### B. Presentation of Reduction Algorithms

In this article, we construct the low-order system using projection-based model reduction. Two main categories of the algorithm are found in the literature: singular value decomposition (SVD)-based methods and Krylov (moment-matching)-based methods. Balanced truncation and proper orthogonal decomposition (POD) are two SVD-based methods. The first one requires the computation of the system's gramian, which can sometimes be computationally expensive, but has the major advantage to offer an *a priori* error bound. The second method, POD, also offers such a bound and is, in addition, directly usable for nonlinear large-scale systems. For moment-matching reduction algorithms, one can find the iterative rational Krylov algorithm [16] and its multi-input-multioutput (MIMO) version [17]. These methods are computationally tractable even for very large-scale systems. In this article, we can use any of the aforementioned reduction methods.

The large-scale state  $x$  can be decomposed into a low-order state  $x_r \in \mathbb{R}^r$  and a neglected state  $x_{\bar{r}} \in \mathbb{R}^{n-r}$  such that

$$x = V_r x_r + V_{\bar{r}} x_{\bar{r}} \quad \text{with} \quad \begin{cases} x_r = W_r^T x \\ x_{\bar{r}} = W_{\bar{r}}^T x \end{cases} \quad (2)$$

The projectors  $W$  and  $V$  are orthogonal, and it holds  $W_r^T V_r = I$  and  $W_r^T V_{\bar{r}} = 0$ . It is now possible to design a controller based on the low-order state  $x_r$ . From (1) and (2), the dynamics of the low-order state writes

$$x_{r+} = \underbrace{W_r^T A V_r}_{A_r} x_r + \underbrace{W_r^T B}_{B_r} u + W_r^T A V_{\bar{r}} x_{\bar{r}}. \quad (3)$$

The dynamics of the low-order state  $x_r$  still depends on the neglected state  $x_{\bar{r}}$  and the matrix  $W_r^T A V_{\bar{r}} \in \mathbb{R}^{r \times (n-r)}$  is of large dimensions. Of course, the controller designed should ensure stability of the exact low-order model (3), and the states  $x_{\bar{r}}$  have to be taken into account. Considering that the term based on the unknown state  $x_{\bar{r}}$  should be modest compared with  $x_r$  (due to the model reduction), a way to tackle this issue is to consider the error reduction as an unknown disturbance  $d_u$  associated with an unavoidable modeling error  $\varepsilon$ . Thus, an adequate controller has to cope with these uncertainties to make them vanish via the robustness property of the feedback control law.

### C. Contributions

This article is a very important extension to our previous work [15] that presented an FEM-based dynamic controller for soft robots. Especially, it introduces dynamic trajectory tracking by modifying the structure of the control law, and it presents new synthesis via LMI constraints' problems to set the gains of both control law and observer. Moreover, a stability proof of the observer-based control is provided, including a guaranteed region of convergence via input to state stability (ISS) properties.

In the rest of this article, we present a trajectory tracking controller based on a low-order model with uncertainties [see model  $M_r$ , (8)]. The control strategy is presented in Sections IV and V, and it includes both feed-forward and observer-based output feedback elements. Under the assumption made, the proposed method ensures the stability of the full-order model  $M$  [see (1)] through LMI problems. The algorithm, computed thanks to the low-order model, is then applied to the full-order model and real-world experiments are gathered in Section VI.

### III. LOW-ORDER MODEL

As said earlier, (3) presents the exact dynamics of the low-order state  $x_r$ , but it still depends on the neglected state  $x_{\bar{r}}$  that is of large dimensions. To ensure the stability of the exact reduced-order model (3), we decompose the reduction error  $W_r^T A V_{\bar{r}} x_{\bar{r}}$  into two orthogonal parts: an input disturbance  $d_u$  plus an unavoidable modeling error  $\varepsilon$

$$W_r^T A V_{\bar{r}} x_{\bar{r}} = B_r d_u + B_r^\perp \varepsilon = \begin{pmatrix} B_r & B_r^\perp \end{pmatrix} \begin{pmatrix} d_u \\ \varepsilon \end{pmatrix} \quad (4)$$

where  $B_r^\perp$  is an orthogonal complement of  $B_r$ , such as  $B_r^{\perp T} B_r = 0$  and  $B_r^{\perp T} B_r^\perp = I$  and the square matrix  $\begin{pmatrix} B_r & B_r^\perp \end{pmatrix}$  is full rank.

*Lemma 1:* Assuming that  $x_{\bar{r}}$  is bounded in a domain around the trajectory, then a bound of  $\varepsilon$  is given by the following equation:

$$\|\varepsilon\| \leq \eta \quad (5)$$

and a overestimated upper bound  $\eta$  of the modeling error  $\varepsilon$  is

$$\begin{aligned} W_r^T A V_{\bar{r}} x_{\bar{r}} &= B_r d_u + B_r^\perp \varepsilon \\ B_r^{\perp T} W_r^T A V_{\bar{r}} x_{\bar{r}} &= B_r^{\perp T} B_r d_u + B_r^{\perp T} B_r^\perp \varepsilon \\ B_r^{\perp T} W_r^T A V_{\bar{r}} x_{\bar{r}} &= \varepsilon \\ \Rightarrow \|\varepsilon\| &\leq \|B_r^{\perp T} W_r^T A V_{\bar{r}} x_{\bar{r}}\| = \eta. \end{aligned} \quad (6)$$

The study of the reduction error  $W_r^T A V_{\bar{r}} x_{\bar{r}}$  gives a value of this bound. Not all reduction methods listed earlier provide the computation of the large matrix  $V_{\bar{r}}$ , as it is computationally expensive, but one can study this bound using

$$\begin{aligned} x_+ &= Ax + Bu \\ W_r^T x_+ &= W_r^T Ax + W_r^T Bu \\ W_r^T x_+ &= W_r^T A(V_r x_r + V_{\bar{r}} x_{\bar{r}}) + W_r^T Bu \\ W_r^T x_+ &= W_r^T A V_r W_r^T x + W_r^T A V_{\bar{r}} x_{\bar{r}} + W_r^T Bu \\ \Rightarrow W_r^T A V_{\bar{r}} x_{\bar{r}} &= W_r^T x_+ - A_r W_r^T x - B_r u. \end{aligned} \quad (7)$$

It is always possible to get an estimation of the upper bound of  $\varepsilon$ . For a simple trajectory, running simulations around the trajectory is sufficient. For a global bound, simulations can be provided covering the entire workspace of the effector and considering the worst case. Experimental values of this bound are provided in Section VI with the experimental validation of the method.

Finally, with notations presented above and from (3) and (4), the low-order dynamics writes

$$M_r : \begin{cases} x_{r+} = A_r x_r + B_r u + B_r d_u + B_r^\perp \varepsilon \\ y_r = C_r x_r. \end{cases} \quad (8)$$

### IV. CONTROL DESIGN

#### A. Reference Model

The trajectory is defined through a linear reference model  $M^*$  sharing the same dimensions as the low-order model  $M_r$ , and  $s^*$  is the reference signal

$$M^* : \begin{cases} x_{r+}^* = A_r^* x_r^* + B_r^* s^* \\ y_r^* = C_r^* x_r^*. \end{cases} \quad (9)$$

The trajectory error  $e_t$  is defined in the reduced-order state space as

$$e_t = x_r - x_r^*. \quad (10)$$

The objective is to design a controller such that it minimizes the tracking error  $e_t$ . To achieve this objective, an observer-based output feedback controller is designed that uses both feed-forward and feedback elements.

#### B. Observer design

The design is based on a so-called unknown input PI-observer. It assumes that the dynamics of  $d_u$  can be captured via a cascade of integrators, i.e., the  $p$ th variation of  $d_u$  is zero. Under this assumption, denoting  $\tilde{B}_r = \begin{pmatrix} B_r & 0 \dots 0 \end{pmatrix}$  and defining an extended state  $(\hat{x}_r^T \hat{d}_u^T)^T$ , the observer writes

$$\begin{pmatrix} \hat{x}_{r+} \\ \hat{d}_{u+} \end{pmatrix} = \begin{pmatrix} A_r & \tilde{B}_r \\ 0 & J \end{pmatrix} \begin{pmatrix} \hat{x}_r \\ \hat{d}_u \end{pmatrix} - \mathbf{K}_o C_r (x_r - \hat{x}_r) + \begin{pmatrix} B_r \\ 0 \end{pmatrix} u \quad (11)$$

with

$$\underbrace{\begin{pmatrix} \hat{d}_{u+} \\ \hat{d}_{u+}^{(1)} \\ \vdots \\ \hat{d}_{u+}^{(p-1)} \\ \hat{d}_{u+}^{(p)} \end{pmatrix}}_{\hat{D}_{u+}} = \underbrace{\begin{pmatrix} I & I & 0 & \dots & 0 \\ 0 & I & I & \ddots & 0 \\ \vdots & 0 & \ddots & \ddots & 0 \\ \vdots & \vdots & \ddots & I & I \\ 0 & 0 & \dots & 0 & I \end{pmatrix}}_J \underbrace{\begin{pmatrix} \hat{d}_u \\ \hat{d}_u^{(1)} \\ \vdots \\ \hat{d}_u^{(p-1)} \\ \hat{d}_u^{(p)} \end{pmatrix}}_{\hat{D}_u}. \quad (12)$$

To keep the observability property, the following constraint must be satisfied:

$$\text{rank}(B_r) \leq p \quad (13)$$



where  $p$  is the number of outputs. Let us define the observation error  $e_o$

$$\begin{aligned} e_o &= \begin{pmatrix} x_r - \hat{x}_r \\ D_u - \hat{D}_u \end{pmatrix} = \begin{pmatrix} e_{o_x} \\ e_{o_d} \end{pmatrix} \\ \Rightarrow e_{o+} &= \underbrace{\begin{bmatrix} A_r & \tilde{\mathbf{B}}_r \\ 0 & J \end{bmatrix} - \mathbf{K}_0(C_r \ O)}_{\mathbf{A}_0 - \mathbf{K}_0 \mathbf{C}_0} e_o + \begin{pmatrix} B_r^\perp \\ 0 \end{pmatrix} \varepsilon \end{aligned} \quad (14)$$

The computation of the matrix  $\mathbf{K}_0$  is detailed in Section V.

### C. Observer-Based Output Feedback

The control law used to perform dynamic trajectory tracking is a PI-like control ( $L_i$ , integral part) based on a reference model state  $x_r^*$  ( $L^*$ ) and the estimated states  $\hat{x}_r$ ,  $\hat{d}_u$  ( $L$  and  $L_u$ )

$$u = -L^* x_r^* - \begin{pmatrix} L & L_u & L_i \end{pmatrix} \begin{pmatrix} \hat{x}_r \\ \hat{d}_u \\ x_i \end{pmatrix}. \quad (15)$$

The dynamics of the trajectory error  $e_t$  writes, from (9) and (8)

$$\begin{aligned} e_{t+} &= x_{r+} - x_{r+}^* \\ &= A_r x_r + B_r u + B_r d_u + \varepsilon - A_r^* x_r^* - B_r^* s^*. \end{aligned} \quad (16)$$

The proposed control law leads to the following trajectory error dynamics:

$$\begin{aligned} e_{t+} &= A_r x_r - B_r (L \hat{x}_r + L^* x_r^* + L_u \hat{d}_u + L_i x_i) + B_r d_u + B_r^\perp \varepsilon \\ &\quad - A_r^* x_r^* - B_r^* s^* \\ e_{t+} &= A_r (e_t + x_r^*) - B_r (L \hat{x}_r + L^* x_r^* + L_u \hat{d}_u + L_i x_i) \\ &\quad + B_r d_u + B_r^\perp \varepsilon - A_r^* x_r^* - B_r^* s^*. \end{aligned} \quad (17)$$

As  $\hat{x}_r = x_r - e_o = e_t + x_r^* - e_o$ , it holds

$$\begin{aligned} e_{t+} &= A_r (e_t + x_r^*) \\ &\quad - B_r (L (e_t + x_r^* - e_{o_x}) + L^* x_r^* + L_u \hat{d}_u + L_i x_i) \\ &\quad + B_r d_u + B_r^\perp \varepsilon - A_r^* x_r^* - B_r^* s^* \\ e_{t+} &= (A_r - B_r L) e_t + (A_r - A_r^* - B_r (L + L^*)) x_r^* \\ &\quad + B_r L e_{o_x} + B_r (d_u - L_u \hat{d}_u) - B_r L_i x_i - B_r^* s^* + B_r^\perp \varepsilon. \end{aligned} \quad (18)$$

In (18), an adequate choice to recover  $d_u - \hat{d}_u$  is to set  $L_u = I$  to get

$$\begin{aligned} e_{t+} &= (A_r - B_r L) e_t + (A_r - A_r^* - B_r (L + L^*)) x_r^* \\ &\quad + (B_r L \quad B_r) \begin{pmatrix} e_{o_x} \\ d_u - \hat{d}_u \end{pmatrix} - B_r L_i x_i - B_r^* s^* + B_r^\perp \varepsilon \\ e_{t+} &= (A_r - B_r L) e_t + (A_r - A_r^* - B_r (L + L^*)) x_r^* \\ &\quad + (B_r L \quad \tilde{\mathbf{B}}_r) e_o - B_r L_i x_i - B_r^* s^* + B_r^\perp \varepsilon. \end{aligned} \quad (19)$$

The integral term  $x_i$  is defined as

$$\begin{aligned} x_{i+} &= x_i + (y_r - y_r^*) \\ x_{i+} &= x_i + C_r e_t + (C_r - C_r^*) x_r^*. \end{aligned} \quad (20)$$

Defining  $\mathbf{z} = \begin{pmatrix} e_t \\ x_r^* \\ x_i \end{pmatrix}$  and  $\mathbf{A}_c = \begin{pmatrix} A_r & A_r - A_r^* & 0 \\ 0 & A_r^* & 0 \\ C_r & C_r - C_r^* & I \end{pmatrix}$ , it follows:

$$\begin{aligned} \mathbf{z}_+ &= \left[ \mathbf{A}_c - \begin{pmatrix} B_r \\ 0 \\ 0 \end{pmatrix} \begin{pmatrix} L & (L + L^*) & L_i \end{pmatrix} \right] \mathbf{z} \\ &\quad + \begin{pmatrix} B_r L & \tilde{\mathbf{B}}_r \\ 0 & 0 \\ 0 & 0 \end{pmatrix} e_o + \begin{pmatrix} -B_r^* \\ B_r^* \\ 0 \end{pmatrix} s^* + \begin{pmatrix} B_r^\perp \\ 0 \\ 0 \end{pmatrix} \varepsilon. \end{aligned} \quad (21)$$

Now, we can write the closed-loop model, including the observer and the controller as

$$\begin{pmatrix} \mathbf{z}_+ \\ e_{o+} \end{pmatrix} = \mathbf{G} \begin{pmatrix} \mathbf{z} \\ e_o \end{pmatrix} + \tilde{\mathbf{B}}_r^* s^* + \tilde{\Phi} \varepsilon \quad (22)$$

with

$$\mathbf{G} = \begin{pmatrix} \mathbf{A}_c - \begin{pmatrix} B_r \\ 0 \\ 0 \end{pmatrix} \begin{pmatrix} L & (L + L^*) & L_i \end{pmatrix} & \begin{pmatrix} B_r L & \tilde{\mathbf{B}}_r \\ 0 & 0 \\ 0 & 0 \end{pmatrix} \\ 0 & \mathbf{A}_0 - \mathbf{K}_0 \mathbf{C}_0 \end{pmatrix} \quad (23)$$

and

$$\tilde{\mathbf{B}}_r^* = \begin{pmatrix} \begin{bmatrix} -B_r^* \\ B_r^* \\ 0 \\ 0 \end{bmatrix} \end{pmatrix}; \quad \tilde{\Phi} = \begin{pmatrix} \begin{bmatrix} B_r^\perp \\ 0 \\ 0 \\ B_r^\perp \\ 0 \end{bmatrix} \end{pmatrix}. \quad (24)$$

Section V present how to compute the observer and controller gains that, under the assumption mentioned in Lemma 1, stabilize the closed-loop system and minimize the trajectory error. In the general case, finding all-in-one the Lyapunov function, the observer, and controller gains result is a non-convex problem. The structure of (21) associated with the matrix  $\mathbf{G}$  of (22) allows to combine a quasi-separation principle (solving the observer design apart from the controller design) and ISS properties to solve the problem. Global stability of the closed loop together with the region of convergence is provided. The control algorithm can thus be summarized as follows.

*Step 1:* Computation of the observer gain.

*Step 2:* Computation of the controller gains.

*Step 3:* Proof of stability and study of convergence region for the closed-loop algorithm [see system (22)].

## V. COMPUTATION OF CONTROLLER AND OBSERVER GAINS

Let us recall the definition of ISS.

[18, Th. 5]: A system with state  $x$  and input  $u$  is ISS if and only if it admits a smooth Lyapunov function  $V(x)$  such that its variation satisfies

$$\Delta V(x, u) \leq -\alpha(\|x\|) + \gamma(\|u\|) \quad (25)$$

where  $\alpha, \gamma \in \mathcal{K}_\infty$ .  $\square$

In the rest of this article, we will use quadratic  $\mathcal{K}_\infty$  functions defined as

$$\alpha(\|x\|) = \alpha x^T M x, \quad \alpha > 0, \quad M > 0. \quad (26)$$

The ISS property will come at hand to study the stability property of the closed loop according to both the modeling error  $\epsilon$  and the input disturbance  $d_u$ .

#### A. Computation of Observer Gain

The first objective is to compute  $\mathbf{K}_o$  in (14) while minimizing the impact of the unknown modeling error  $\epsilon$  on the observer error  $e_o$  that writes

Find  $\mathbf{K}_o$  and  $V_o(e_o) = e_o^T P_o e_o$  for the model:

$$e_{o+} = (\mathbf{A}_o - \mathbf{K}_o \mathbf{C}_o) e_o + \begin{pmatrix} B_r^\perp \\ 0 \end{pmatrix} \epsilon$$

such that:

$$\Delta V_o < -\alpha_o e_o^T P_o e_o + \gamma_o \epsilon^T \epsilon$$

with:  $\alpha_o > 0$ ,  $\gamma_o > 0$ ,  $P_o > 0$ .

(27)

(27) is equivalent to

$$\begin{aligned} & [*] P_o \left[ (\mathbf{A}_o - \mathbf{K}_o \mathbf{C}_o) e_o + \begin{pmatrix} B_r^\perp \\ 0 \end{pmatrix} \epsilon \right] \\ & - e_o P_o e_o < -\alpha_o e_o^T P_o e_o + \gamma_o \epsilon^T \epsilon \\ \Leftrightarrow & (e_o^T \quad \epsilon^T) [*] P_o \begin{bmatrix} \mathbf{A}_o - \mathbf{K}_o \mathbf{C}_o & \begin{pmatrix} B_r^\perp \\ 0 \end{pmatrix} \end{bmatrix} \begin{pmatrix} e_o \\ \epsilon \end{pmatrix} \\ & + (e_o^T \quad \epsilon^T) \begin{pmatrix} -P_o + \alpha_o P_o & 0 \\ 0 & -\gamma_o I \end{pmatrix} \begin{pmatrix} e_o \\ \epsilon \end{pmatrix} < 0. \end{aligned} \quad (28)$$

Using Schur's complement, this is equivalent to the following matrix inequality condition:

$$(27) \Leftrightarrow \begin{pmatrix} -P_o + \alpha_o P_o & 0 & * \\ 0 & -\gamma_o I & * \\ P_o \mathbf{A}_o - \tilde{\mathbf{K}}_o \mathbf{C}_o & P_o \begin{bmatrix} B_r^\perp \\ 0 \end{bmatrix} & -P_o \end{pmatrix} < 0 \quad (29)$$

with the classical change of variables  $\tilde{\mathbf{K}}_o = P_o \mathbf{K}_o$ . It ensures that for large enough time, the trajectory error converges to a hyperball, whose radius depends on  $\gamma_o$  and  $\alpha_o$ .

#### B. Computation of Controller Gain

The second objective is to compute the controller gains  $L$ ,  $L^*$ , and  $L_i$  in (15), while minimizing the impact of the modeling and reduction errors on the output tracking error  $(y - y^*)$

Find  $L$ ,  $L^*$ ,  $L_i$  and  $V_t(\mathbf{z}) = \mathbf{z}^T P_t \mathbf{z}$  for the model:

$$\begin{aligned} \mathbf{z}_+ = & \begin{bmatrix} \mathbf{A}_c - \begin{pmatrix} B_r \\ 0 \\ 0 \end{pmatrix} (L \quad (L + L^*) \quad L_i) \end{bmatrix} \mathbf{z} \\ & + \begin{pmatrix} B_r L \quad \tilde{\mathbf{B}}_r \\ 0 \quad 0 \\ 0 \quad 0 \end{pmatrix} e_o + \begin{pmatrix} -B_r^* \\ B_r^* \\ 0 \end{pmatrix} s^* + \begin{pmatrix} B_r^\perp \\ 0 \\ 0 \end{pmatrix} \epsilon \end{aligned} \quad (30)$$

such that:

$$\Delta V_t < -\alpha(y - y^*)^T (y - y^*) + \beta_t e_o^T \begin{pmatrix} P_t & 0 \\ 0 & I \end{pmatrix} e_o$$

$$+ \gamma_t \epsilon^T \epsilon + \omega_t s^{*T} s^*$$

with:  $(\alpha_t, \beta_t, \gamma_t, \omega_t) > 0$  and  $P_t > 0$ .

It holds

$$(y - y^*) = (C_r \quad C_r - C_r^* \quad 0) \begin{pmatrix} e_t \\ x_r^* \\ x_i \end{pmatrix} = \tilde{\mathbf{C}}_r \mathbf{z}$$

$$\Rightarrow (y - y^*)^T (y - y^*) = \mathbf{z}^T \tilde{\mathbf{C}}_r^T \tilde{\mathbf{C}}_r \mathbf{z}. \quad (31)$$

Thus, problem (31) is equivalent to

$$\begin{aligned} & [*] P_t \left[ \begin{pmatrix} \mathbf{A}_c - \begin{pmatrix} B_r \\ 0 \\ 0 \end{pmatrix} (L \quad (L + L^*) \quad L_i) \end{pmatrix} \mathbf{z} \right. \\ & \quad \left. + \begin{pmatrix} B_r L \quad \tilde{\mathbf{B}}_r \\ 0 \quad 0 \\ 0 \quad 0 \end{pmatrix} e_o + \begin{pmatrix} -B_r^* \\ B_r^* \\ 0 \end{pmatrix} s^* + \begin{pmatrix} B_r^\perp \\ 0 \\ 0 \end{pmatrix} \epsilon \right] - \mathbf{z}^T P_t \mathbf{z} \\ & < -\alpha_t \mathbf{z}^T \tilde{\mathbf{C}}_r^T \tilde{\mathbf{C}}_r \mathbf{z} + \beta_t e_o^T \begin{pmatrix} P_t & 0 \\ 0 & I \end{pmatrix} e_o + \gamma_t \epsilon^T \epsilon + \omega_t s^{*T} s^*. \end{aligned} \quad (32)$$

Denoting  $X_t = P_t^{-1}$  and using Schur's complement and congruence property with  $\mathbf{diag}(X_t, [X_t \quad I], I, I, I)$ , condition (33) shown in bottom of the next page, defines an LMI problem corresponding to the ISS condition (31). LMI problems are solved using YALMIP for MATLAB [19].

#### C. Proof of Stability for Complete Low-Order System

The two previous LMI problems compute the controller and observer gains. This section optimizes the convergence region, where the proof of stability is given. Once the observer and controllers gains are obtained, the matrix  $\mathbf{G}$  defined in (23) is fully defined. The stability of the complete closed loop (i.e., the controller and observer) is guaranteed if the following problem is solved.

System (22) is ISS if it exists a Lyapunov function  $V(\mathbf{z}, e_o) = (\mathbf{z}^T \quad e_o^T) P \begin{pmatrix} \mathbf{z} \\ e_o \end{pmatrix}$  such that

$$\Delta V(\mathbf{z}, e_o) < -\alpha(\|\mathbf{z}\| \quad \|e_o\|) + \gamma(\|\epsilon\|) + \omega(\|s^*\|). \quad (34)$$

We define  $\alpha(\|x\|) = \alpha x^T P x$ ,  $\omega(\|s^*\|) = \omega s^{*T} s^*$ , and  $\gamma(\|\epsilon\|) = \gamma \epsilon^T \epsilon$  with  $P > 0$ ,  $\alpha > 0$ ,  $\omega > 0$ , and  $\gamma > 0$ . Then, the previous ISS condition is satisfied if

$$\begin{aligned} (34) \Leftrightarrow & [*] P \left[ \mathbf{G} \begin{pmatrix} \mathbf{z} \\ e_o \end{pmatrix} + \tilde{\mathbf{B}}_r^* s^* + \tilde{\Phi} \epsilon \right] - \begin{pmatrix} \mathbf{z} \\ e_o \end{pmatrix}^T P \begin{pmatrix} \mathbf{z} \\ e_o \end{pmatrix} \\ & < -\alpha \begin{pmatrix} \mathbf{z} \\ e_o \end{pmatrix}^T P \begin{pmatrix} \mathbf{z} \\ e_o \end{pmatrix} + \gamma \epsilon^T \epsilon + \omega s^{*T} s^* \\ \Leftrightarrow & [*] P [G \quad \tilde{\mathbf{B}}_r^* \quad \tilde{\Phi}] - \mathbf{diag}((1 - \alpha)P, \omega I, \gamma I) < 0. \end{aligned} \quad (35)$$

This defines a generalized eigenvalue problem in  $P, \alpha, \gamma$ , and  $\omega$  and ensures the stability of system (34) with the observer gain  $\mathbf{K}_o$  and controller gains  $L$ ,  $L^*$ , and  $L_i$ . Moreover, it ensures that  $V(\mathbf{z}, e_o)$  is decreasing for large enough vectors  $(\mathbf{z}, e_o)$ . The system finally converges to an invariant manifold defined as

$$\begin{pmatrix} \mathbf{z} \\ e_o \end{pmatrix}^T P \begin{pmatrix} \mathbf{z} \\ e_o \end{pmatrix} < \frac{\gamma \eta^2 + \omega \|s^*\|^2}{\alpha}. \quad (36)$$

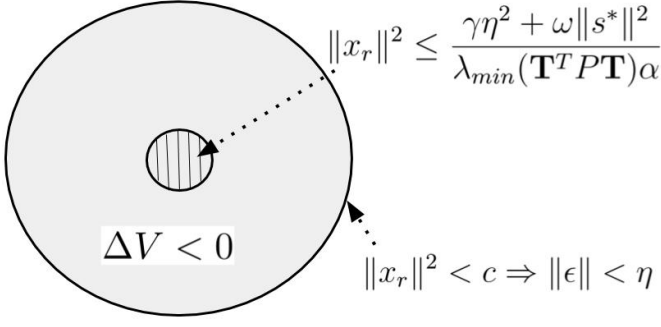


Fig. 1. Illustration of the convergence region using the ISS property. The hatching circle corresponds to invariant manifold defined in (41).

This defines a region of convergence for the vector  $(\mathbf{z}, e_o)$ . In order to study the convergence of the reduced-order state  $x_r$ , let us write

$$\begin{pmatrix} \mathbf{z} \\ e_o \end{pmatrix} = \begin{pmatrix} e_t \\ x_r^* \\ x_i \\ e_o \end{pmatrix} = \begin{pmatrix} I & -I & 0 & 0 \\ 0 & I & 0 & 0 \\ 0 & 0 & I & 0 \\ 0 & 0 & 0 & I \end{pmatrix} \begin{pmatrix} x_r \\ x_r^* \\ x_i \\ e_o \end{pmatrix} = \mathbf{T} \begin{pmatrix} x_r \\ x_r^* \\ x_i \\ e_o \end{pmatrix}. \quad (37)$$

Equation (36) is also equivalent to

$$\begin{pmatrix} \mathbf{z} \\ e_o \end{pmatrix}^T P \begin{pmatrix} \mathbf{z} \\ e_o \end{pmatrix} = \begin{pmatrix} x_r \\ x_r^* \\ x_i \\ e_o \end{pmatrix}^T \mathbf{T}^T P \mathbf{T} \begin{pmatrix} x_r \\ x_r^* \\ x_i \\ e_o \end{pmatrix} < \frac{\gamma \eta^2 + \omega \|s^*\|^2}{\alpha}. \quad (38)$$

In addition, we have

$$\begin{pmatrix} x_r \\ x_r^* \\ x_i \\ e_o \end{pmatrix}^T \mathbf{T}^T P \mathbf{T} \begin{pmatrix} x_r \\ x_r^* \\ x_i \\ e_o \end{pmatrix} \geq \begin{pmatrix} x_r \\ x_r^* \\ x_i \\ e_o \end{pmatrix}^T \begin{pmatrix} x_r \\ x_r^* \\ x_i \\ e_o \end{pmatrix} \lambda_{\min}(\mathbf{T}^T P \mathbf{T}) \quad (39)$$

where  $\lambda_{\min}(A)$  is the smallest eigenvalue of  $A$ . It yields

$$\begin{pmatrix} x_r \\ x_r^* \\ x_i \\ e_o \end{pmatrix}^T \mathbf{T}^T P \mathbf{T} \begin{pmatrix} x_r \\ x_r^* \\ x_i \\ e_o \end{pmatrix} \geq x_r^T x_r \lambda_{\min}(\mathbf{T}^T P \mathbf{T}). \quad (40)$$

Finally, the invariant manifold with respect to the reduced-order state writes

$$\|x_r\|^2 \leq \frac{\gamma \eta^2 + \omega \|s^*\|^2}{\lambda_{\min}(\mathbf{T}^T P \mathbf{T})\alpha}. \quad (41)$$

A graphical scheme of the convergence region defined by (41) is shown in Fig. 1. Section VI will give an example of

the estimation of this bound as well as the experimental results from the studied robot.

#### D. Summary of Closed-Loop Algorithm

The study of the open-loop system says that for a given workspace in which the reduced-order state  $x_r$  is bounded by a given constant  $c$ , the modeling error  $\epsilon$  is bounded by  $\eta$

$$\|x_r\|^2 < c \Rightarrow \|\epsilon\| < \eta.$$

Then, three LMI problems guarantee that if the modeling error  $\epsilon$  is bounded by  $\eta$ , then the reduced-order state converges to an invariant manifold defined in (36) and (41). If this region of convergence is smaller than the workspace in which  $\|x_r\|^2 < c$  is satisfied, then the closed-loop system is stable.

##### 1) Control Design Algorithm:

Study of open-loop system

$$\|\epsilon\| < \eta \Rightarrow \|x_r\|^2 \leq \frac{\gamma \eta^2 + \omega \|s^*\|^2}{\lambda_{\min}(\mathbf{T}^T P \mathbf{T})\alpha}.$$

Solve three LMI problems.

$$\text{If } \frac{\gamma \eta^2 + \omega \|s^*\|^2}{\lambda_{\min}(\mathbf{T}^T P \mathbf{T})\alpha} < c$$

then the closed-loop is stable.

#### VI. EXPERIMENTAL VALIDATION

To illustrate the interest and the effectiveness of the methodology proposed, it is tested on soft robotics application. The modeling step is done using the SOFA framework [12], a unified software to model soft structures thanks to the FEM. For this model to be precise, the number of nodes of the mesh has to be significantly high and the model studied is also a large-scale system.

##### A. Experimental Setup

Experiments are conducted on a trunk-like robot presented in Fig. 2 with a schematic view in Fig. 3. It is entirely made of silicone, the structure is 18 cm long, and the thicknesses at its base and its tip are, respectively, 2.5 and 1 cm. The structure is driven by four cables—actuated by four servomotors—that permit to work in the three dimensions of space. The output of the system is the position of the tip (red point in Fig. 2) that is measured using a magnetic microsensor, whose frequency can reach 240 Hz. In the following, the sampling time for the real-time experiments is set to 25 Hz.

$$\begin{pmatrix} -X_t & * & * & * & * \\ 0 & -\beta_t \begin{pmatrix} X_t & 0 \\ 0 & I \end{pmatrix} & * & * & * \\ 0 & 0 & -\omega_t I & * & * \\ 0 & 0 & 0 & -\gamma_t I & * \\ \mathbf{A}_c - \begin{pmatrix} B_r \\ 0 \\ 0 \end{pmatrix} (L \quad (L + L^*) \quad L_i) X_t & \begin{pmatrix} B_r L X_t & \tilde{\mathbf{B}}_r \\ 0 & 0 \\ 0 & 0 \end{pmatrix} & \begin{pmatrix} -B_r^* \\ B_r^* \\ 0 \end{pmatrix} & \begin{pmatrix} B_r^\perp \\ 0 \\ 0 \end{pmatrix} & -X_t \quad * \\ \tilde{\mathbf{C}}_r X_t & 0 & 0 & 0 & -\alpha_t I \end{pmatrix} < 0 \quad (33)$$



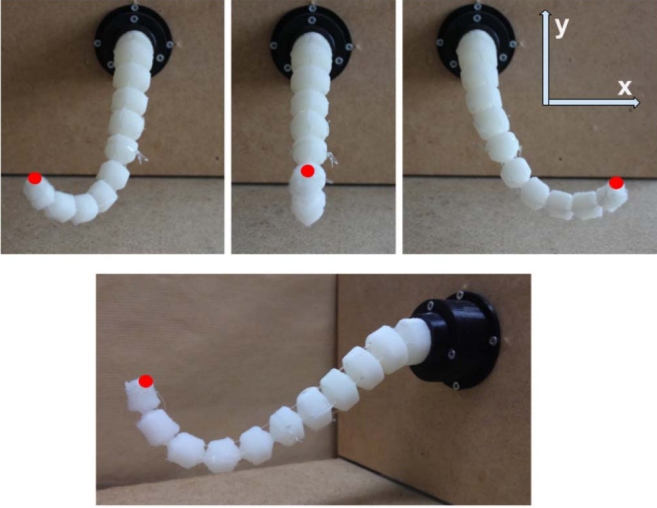


Fig. 2. Soft robot used for experimental validation. It is entirely made of silicone. Top: front view of the robot. Bottom: side view. Red = end-effector, location of the sensor.

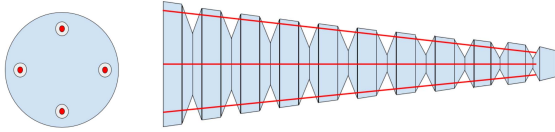


Fig. 3. Design of the robot. Slice view (left) and side view (right). The robot is actuated with four cables in red.

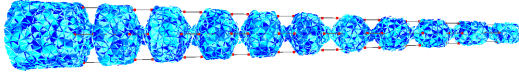


Fig. 4. FEM mesh of the trunk-like robot presented in Fig. 2. The mesh is made of 1557 nodes and 2972 tetrahedrons.

### B. Modeling

The finite-element mesh of this robot is made of 1557 nodes, see Fig. 4. The dimensions of the state vector in system (55) is also  $1557 \times 3 \times 2 = 9342$  state variables (three directions of space for displacement and velocity). Then, model-order reduction provides us with a low-order system of dimension 6, as explained in Section VI-D.

1) *Nonlinear Large-Scale Model*: Let us start with the formulation given by the second law of Newton that models the dynamic behavior of a body as

$$\mathbb{M}(q)\dot{v} = \mathbb{P}(q) - \mathbb{F}(q, v) + \mathbb{H}^T(q)\lambda \quad (42)$$

where  $q \in \mathbb{R}^n$  is the vector of generalized degrees of freedom,  $\mathbb{M}(q)$  is the inertia matrix, and  $v = \dot{q} \in \mathbb{R}^n$  is the vector of velocity.  $\mathbb{F}(q, v)$  represents internal forces applied to the simulated object depending on the current state and  $\mathbb{P}(q)$  gathers known external forces. The matrices  $\mathbb{M}$ ,  $\mathbb{F}$ , and  $\mathbb{P}$  are square matrices of dimensions  $n \times n$ .  $\mathbb{H}^T(q)$  is the matrix containing the constraint directions, while  $\lambda \in \mathbb{R}^m$  is the vector of actuators forces. The procedure to get these different terms is given shortly, and the reader can refer to [12] for details. We integrate (42) using a time-stepping implicit scheme (backward Euler) to have unconditional stability. Let us consider the time interval  $[t_k, t_{k+1}]$  whose length is  $h = t_{k+1} - t_k$ .

In practice due to hardware limitation, the time step is equal to 0.04 s. Let us denote  $\delta q = q_+ - q = hv_+$  and  $\delta v = v_+ - v$ ; with these notations, (42) writes

$$\begin{aligned} \mathbb{M}(q_+)\delta v &= h(\mathbb{P}(q_+) - \mathbb{F}(q_+, v_+)) + h\mathbb{H}^T(q_+)\lambda \\ q_+ &= q + hv_+. \end{aligned} \quad (43)$$

The internal forces  $\mathbb{F}$  are a nonlinear function of the positions and the velocities. We then apply a Taylor series expansion to  $\mathbb{F}$  and make the following first-order approximation:

$$\begin{aligned} \mathbb{F}(q_+, v_+) &= \mathbb{F}(q + \delta q, v + \delta v) \\ &\approx \mathbb{F}(q, v) + \frac{\partial \mathbb{F}(q, v)}{\partial q} \delta q + \frac{\partial \mathbb{F}(q, v)}{\partial v} \delta v. \end{aligned} \quad (44)$$

Defining

$$\mathbf{S}(q, v) = \left( \mathbb{M}(q) + h \frac{\partial \mathbb{F}(q, v)}{\partial v} + h^2 \frac{\partial^2 \mathbb{F}(q, v)}{\partial q \partial v} \right). \quad (45)$$

As  $\delta q = hv_+ = h\delta v + hv$ , it follows:

$$\mathbf{S}(q, v)\delta v = h \left( \mathbb{P}(q_+) - \mathbb{F}(q, v) - h \frac{\partial \mathbb{F}(q, v)}{\partial q} v \right) + h\mathbb{H}^T(q)\lambda. \quad (46)$$

And finally

$$\begin{aligned} v_+ &= v + h\mathbf{S}(q, v)^{-1} \left( \mathbb{P}(q_+) - \mathbb{F}(q, v) - h \frac{\partial \mathbb{F}(q, v)}{\partial q} v \right) \\ &\quad + h\mathbf{S}(q, v)^{-1} \mathbb{H}^T(q)\lambda. \end{aligned} \quad (47)$$

Let us define the state vector  $x$  as

$$x = \begin{pmatrix} v \\ q \end{pmatrix}. \quad (48)$$

The nonlinear state-space expression is given by (49), shown in bottom of the next page.

### C. Large-Scale Linear Discrete-Time State-Space Model

Let  $q_0 \in \mathbb{R}^n$  be a stable equilibrium point that could be obtained using an inverse problem. It is induced by the gravity field  $\mathbb{P}(q_0)$  that is considered constant, i.e.,  $\mathbb{P}(q_0) = \mathbb{P}$ , and the actuation input  $\lambda_0$ . The Taylor expansion is made around this configuration  $q_0$  and it holds

$$\frac{\partial \mathbb{F}(q, v)}{\partial q} \bigg|_{\substack{q=q_0 \\ v=0}} = \mathbb{K} \quad \text{and} \quad \frac{\partial \mathbb{F}}{\partial v} = \frac{\partial \mathbb{F}(q, v)}{\partial v} \bigg|_{\substack{q=q_0 \\ v=0}} = \mathbb{D}. \quad (50)$$

Moreover, we consider constant inertia matrix  $\mathbb{M}(q) = \mathbb{M}$ . With these notations, we get

$$\mathbf{S} = (\mathbb{M} + h\mathbb{D} + h^2\mathbb{K}). \quad (51)$$

Defining the new input vector  $u = \lambda - \lambda_0$  and  $\mathbf{q} = q - q_0$ , the linear state-space equation is

$$\begin{pmatrix} I & 0 \\ -h & I \end{pmatrix} x_+ = \begin{pmatrix} v - h\mathbf{S}^{-1}[\mathbb{F} + h\mathbb{K}v] \\ \mathbf{q} \end{pmatrix} + \begin{pmatrix} h\mathbf{S}^{-1}\mathbb{H}^T \\ 0 \end{pmatrix} u \quad (52)$$

From (49) to (52), the value of  $\mathbb{P}$  vanishes as it is included in the equilibrium point.  $\mathbb{F}$  are the internal forces of the previous

time step that can be written as  $\mathbb{F} = \mathbb{K}q$ . Time step  $h$  is a nonnull scalar, and thus, we have

$$\begin{pmatrix} I & 0 \\ -h & I \end{pmatrix}^{-1} = \begin{pmatrix} I & 0 \\ h & I \end{pmatrix}. \quad (53)$$

With these notations, state space (52) writes

$$\begin{aligned} x_+ = & \begin{pmatrix} I & 0 \\ h & I \end{pmatrix} \begin{pmatrix} v - hS^{-1}[\mathbb{K}q + h\mathbb{K}v] \\ \mathbf{q} \end{pmatrix} \\ & + \begin{pmatrix} I & 0 \\ h & I \end{pmatrix} \begin{pmatrix} hS^{-1}\mathbb{H}^T \\ 0 \end{pmatrix} u. \end{aligned} \quad (54)$$

Finally, the state-space equation can be written in the standard form

$$M : \begin{cases} x_+ = \underbrace{\begin{pmatrix} I - h^2S^{-1}\mathbb{K} & -hS^{-1}\mathbb{K} \\ hI - h^3S^{-1}\mathbb{K} & I - h^2S^{-1}\mathbb{K} \end{pmatrix}}_A \begin{pmatrix} v \\ \mathbf{q} \end{pmatrix} \\ \quad + \underbrace{\begin{pmatrix} hS^{-1}\mathbb{H}^T \\ h^2S^{-1}\mathbb{H}^T \end{pmatrix}}_B u \\ y = Cx \end{cases} \quad (55)$$

where  $A \in \mathbb{R}^{2n \times 2n}$ ,  $B \in \mathbb{R}^{2n \times m}$ ,  $m$  being the number of actuators and  $C \in \mathbb{R}^{p \times 2n}$  is a matrix defining the end-effector coordinates, and  $p$  is the number of outputs.

#### D. Low-Order Model

As the computation of  $\mathcal{H}_\infty$  is computationally more expensive, a common way to measure the accuracy of the reduced-order system is to measure the  $\mathcal{H}_2$  norm error between the full- and reduced-order models

$$\|M - M_r\|_{\mathcal{H}_2}. \quad (56)$$

In this article, we use a moment-matching reduction method, implemented within the MORE toolbox [20], to perform the reduction. Table I gathers the comparison of the  $\mathcal{H}_2$  errors for different sizes of reduced systems, provided that the norm of the full-order model  $\|M\|_{\mathcal{H}_2} = 0.42$ . The best result regarding the  $\mathcal{H}_2$  norm error is achieved for reduced-order systems of dimension 26 and higher; they present an error of 24.67% compared with the full-order model. The  $\mathcal{H}_2$  error between the full-order system and the reduced-order system of dimension 2 is 25.4%, which is close to the optimal solution. Indeed, the  $\mathcal{H}_2$  error decreases from a system with dimension 2 to dimension 6 and then converges slowly to the optimal solution. A good compromise between the accuracy and the complexity of the low-order model is also to choose a reduced-order system of dimension 6.

TABLE I  
COMPARISON OF THE  $\mathcal{H}_2$  NORM ERROR BETWEEN THE FULL- AND REDUCED-ORDER SYSTEMS FOR DIFFERENT DIMENSIONS OF REDUCED SYSTEMS

Dimensions of $M_r$	2	4	6	10	26	30	100
$\mathcal{H}_2$ norm error $\ M - M_r\ _{\mathcal{H}_2}$ in %	25.40	24.70	24.69	24.69	24.67	24.67	24.67

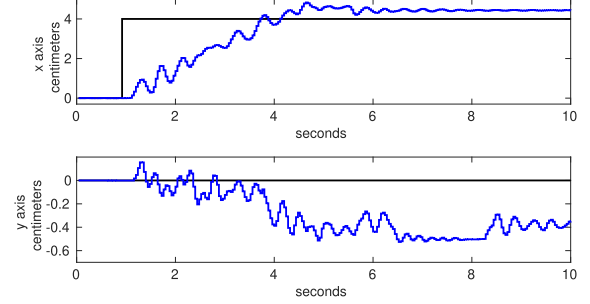


Fig. 5. Real-time open-loop behavior of the robot. Black: reference signal. Blue: end-effector's displacement. Top: measurements of displacement along the x-axis. Bottom: measurements of displacement along the y-axis.

#### E. Real Setup's Behavior in Open Loop

The open-loop behavior of the robot is shown in Fig. 5. It presents the displacements of the end-effector (i.e., the output of the system) in the open loop controlled by solving an inverse problem.

*Remark:* With an accurate calibration process, it is possible to obtain a zero static error in the open loop. However, the open-loop algorithms do not offer any guarantee about the performances of the control algorithm, especially in the presence of perturbations. Moreover, due to hardware limitations, the calibration has to be done regularly, as phenomena, such as fatigue, are difficult to model.

#### F. Validation of Model in Simulations

1) *Validation of Model:* The experiments of Fig. 5 are repeated in simulation, and the results are shown in Fig. 6. Fig. 5 compared with Fig. 6 shows that the main behavior of the robot is captured by the model.

2) *Bound  $\eta$  of the Modeling Error:* By running multiple simulations, we cover a workspace of the robot that is considered as exhaustive (we cover the whole range of possible actuation) and we get the maximal values of the norms of the full- and reduced-order states

$$\|x\| = 2.39 \cdot 10^8 \text{ and } \|x_r\| = 2.37 \cdot 10^8. \quad (57)$$

In this workspace and from (7), the norm of the modeling error is

$$\eta = 4.19 \cdot 10^3. \quad (58)$$

$$\begin{pmatrix} I & 0 \\ -h & I \end{pmatrix} x_+ = \begin{pmatrix} v + hS(q, v)^{-1}(\mathbb{P}(q_+) - \mathbb{F}(q, v) - h \frac{\partial \mathbb{F}(q, v)}{\partial q} v) \\ q \end{pmatrix} + \begin{pmatrix} hS(q, v)^{-1}\mathbb{H}^T(q) \\ 0 \end{pmatrix} \lambda \quad (49)$$

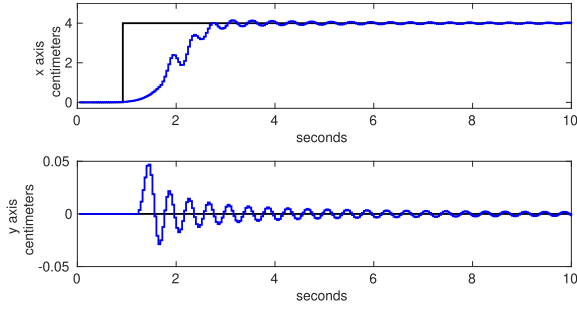


Fig. 6. Open-loop simulation of the robot controlled via an inverse problem. Black: reference signal. Blue: end-effector's displacement. Top: displacement along the x-axis. Bottom: displacement along the y-axis.

3) *Invariant Manifold for the System Studied*: The invariant manifold defined in (41) and depicted in Fig. 1 is defined with the following parameters:

$$\begin{aligned} \alpha &= 0.05 \\ \gamma &= 10^{-2} \\ \omega &= 2.10^{-3}. \end{aligned} \quad (59)$$

For the case where  $s^* = 0$ , i.e., the robot comes back to its rest shape, the invariant manifold writes

$$\|x_r\|^2 \leq \frac{\gamma \eta^2}{\lambda_{\min}(\mathbf{T}^T \mathbf{P} \mathbf{T}) \alpha} = 9.14 \cdot 10^6. \quad (60)$$

In this case, the invariant manifold corresponds to 3.87% of the robot workspace. Therefore, for an initial condition that satisfies (57), the system is stable and converges to the invariant manifold.

### G. Experimental Validation

Experiments are conducted with the same sampling time as for the modeling process; point-to-point control and trajectory tracking experiments are conducted.

1) *Point-to-Point Control*: For the first experiment, the reference is set such that the robot starts from its rest position and converges to a desired position. Once the robot has reached the target, it is moved 1 cm away from the target. Fig. 7 shows that the control algorithm is robust enough to deal with this perturbation. For the second experiment, the robots start from its rest shape and converge to the first deformed position and then to the second deformed position. Results are presented in Fig. 8. It shows a diminution of oscillations, a faster time response, and a cancellation of the static error compared with the open loop.

2) *Sinusoidal Trajectory*: The second set of experiments consists of tracking a sinusoidal signal along the  $x$ -axis of the robot; the reference for  $y$ -axis is zero. Results are presented in Fig. 9. The behavior along the first axis accurately follows the trajectory, while the second output oscillates around the reference signal. Of course, this residual oscillation is due to the physical coupling between the actuators.

### H. Discussion and Limitations

The limits of the algorithm are tested with the third experiment, where the goal is to reach a highly deformed position.

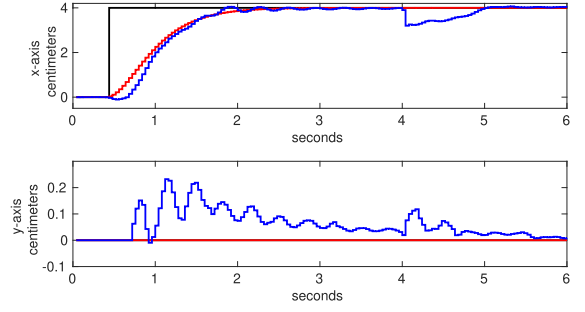


Fig. 7. Point-to-point closed-loop control. Top: output along the  $x$ -axis. Bottom: output along the  $y$ -axis. Black: reference signal. Red: output of reference model. Blue: Robot's end-effector's displacement.

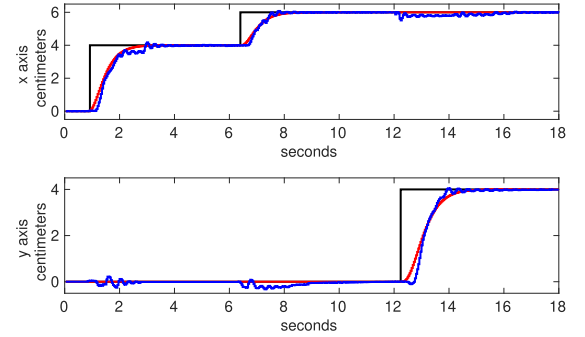


Fig. 8. Point-to-point closed-loop control. Desired position corresponds to Fig. 10(c). Top: output along the  $x$ -axis. Bottom: output along the  $y$ -axis. Black: reference signal. Red: output of the reference model. Blue: robot's end-effector's displacement.

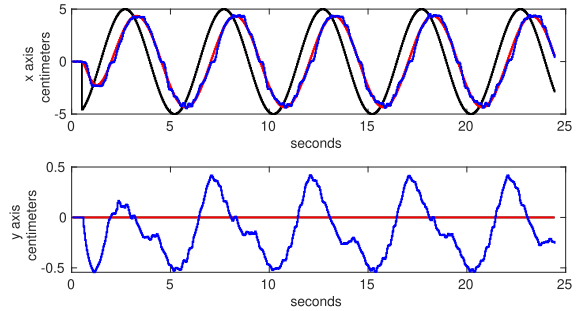


Fig. 9. Closed-loop trajectory tracking. Top: output along the  $x$ -axis. Bottom: output along the  $y$ -axis. Black: reference signal. Red: output of reference model. Blue: robot's end-effector's displacement.

Prior to the experiment, it has been checked via an inverse problem that the target is inside the robot workspace. Fig. 10 shows the different positions, where the algorithm is tested. Fig. 10(a) shows the rest position where the system is linearized. Fig. 10(b) shows a deformation of 5% where the linearization assumption is valid. Fig. 10(c) corresponds to the desired position of the successful attempt shown in Fig. 8. Fig. 10(d) shows an example of unsuccessful attempt to reach a highly deformed position. This limitation is shown in Fig. 11 and is probably due to the fact that the position is outside the validity range of the linearization.

Indeed, the proposed closed-loop controller is based on a linear model: we consider constant mass, stiffness, and damping. This linearization assumption, necessary to cope with the high-sized problem, naturally limits the workspace of the

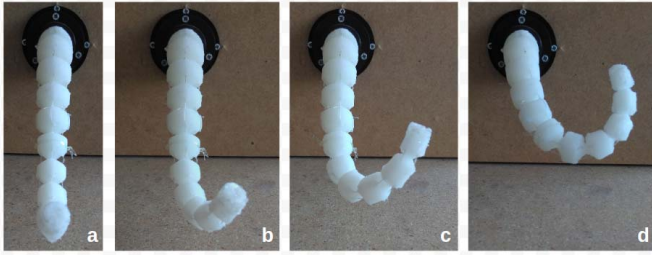


Fig. 10. (a) Rest position. (b) Limits of the linearization assumption (5% of deformation). (c) Desired position corresponding to Fig. 8. (d) Desired position corresponding to Fig. 11.

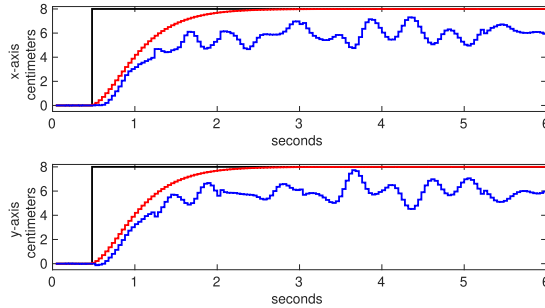


Fig. 11. Unsuccessful attempt to reach a more distant target [see Fig. 10(d)]. Top: output along the  $x$ -axis. Bottom: output along the  $y$ -axis. Black: reference signal. Red: output of reference model. Blue: robot's end-effector's displacement.

robot, and the model remains only valid in a neighborhood of the linearization. In practice, for the robot under consideration, the workspace is limited approximately to a circle of radius 6 cm around its rest position. If the reference trajectory is set outside of this workspace, the performances start to decrease. Even if it is recognized as a limiting factor, this is not a strong limitation as 6 cm represent 33.3% of the length of the structure. Further experimental tests with different robots should be performed to evaluate and characterize the full potential of the approach. For larger deformation, more elaborated models should come at hand (e.g., polytopic descriptions, nonlinear representation, and so on); this is an ongoing research topic.

In addition, the mechanical coupling between the actuators limits the performances of the control algorithm, as shown in Fig. 9. Moreover, the friction between the cables and the structure also limits the performances of the closed-loop algorithm, and this requires deepening the actuation system to remove this constraint.

## VII. CONCLUSION

A large-scale model of a physical system is obtained using the numerical method, such as the FEM. The precision of the model is linked to the size of the model; as accuracy increases, so does the size of the state-space model.

We deal with this dimension issue using projection-based model reduction. The design of a robust controller is done to take into account both the modeling errors and the modeling reduction. A so-called PI-like observer reconstructs the reduction part via an unknown vector. The conditions are written as three LMI constraints' problem. When a solution

to these problems is found, it is applied to the original full-order model.

To illustrate the interest of this article, theoretical results are applied to a soft robot. Dynamic control of soft robots is an open field of research and the results provided in this article show that the method is relevant to increase the accuracy, range of speed, and robustness of this kind of robots.

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